

## Physical phenomena Loads & Failure Mechanisms

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# Outline

- **Background and introduction**
- **Load vs. Capacity**
- **Load types**
- **Failure mechanisms**
- **Applications**
  - Preventive maintenance and prognostics
  - RAMS analysis → Relevant Failure Parameter
  - Condition monitoring
  - Root cause analysis

# Objective

- **Demonstrate benefit of understanding failures**
  - Enables quantification of maintenance
  - Provides structured way of problem solving
- **Give a flavour of loads & failure mechanisms**
  - Not complete overview
  - Simple examples
    - › not representative for your practical problems
    - › illustrate principles
    - › demonstrate quantification – some equations !
- **Provide directions for application**
  - Not complete overview
  - Not in-depth treatment of one case

# Background

## Defence organization:

- Technologically sophisticated systems
- Largely variable and demanding operating conditions
- High requirements for availability and reliability
- Constant pressure to reduce costs



**Maintenance important**

***Predictability of maintenance desired***



# Introduction

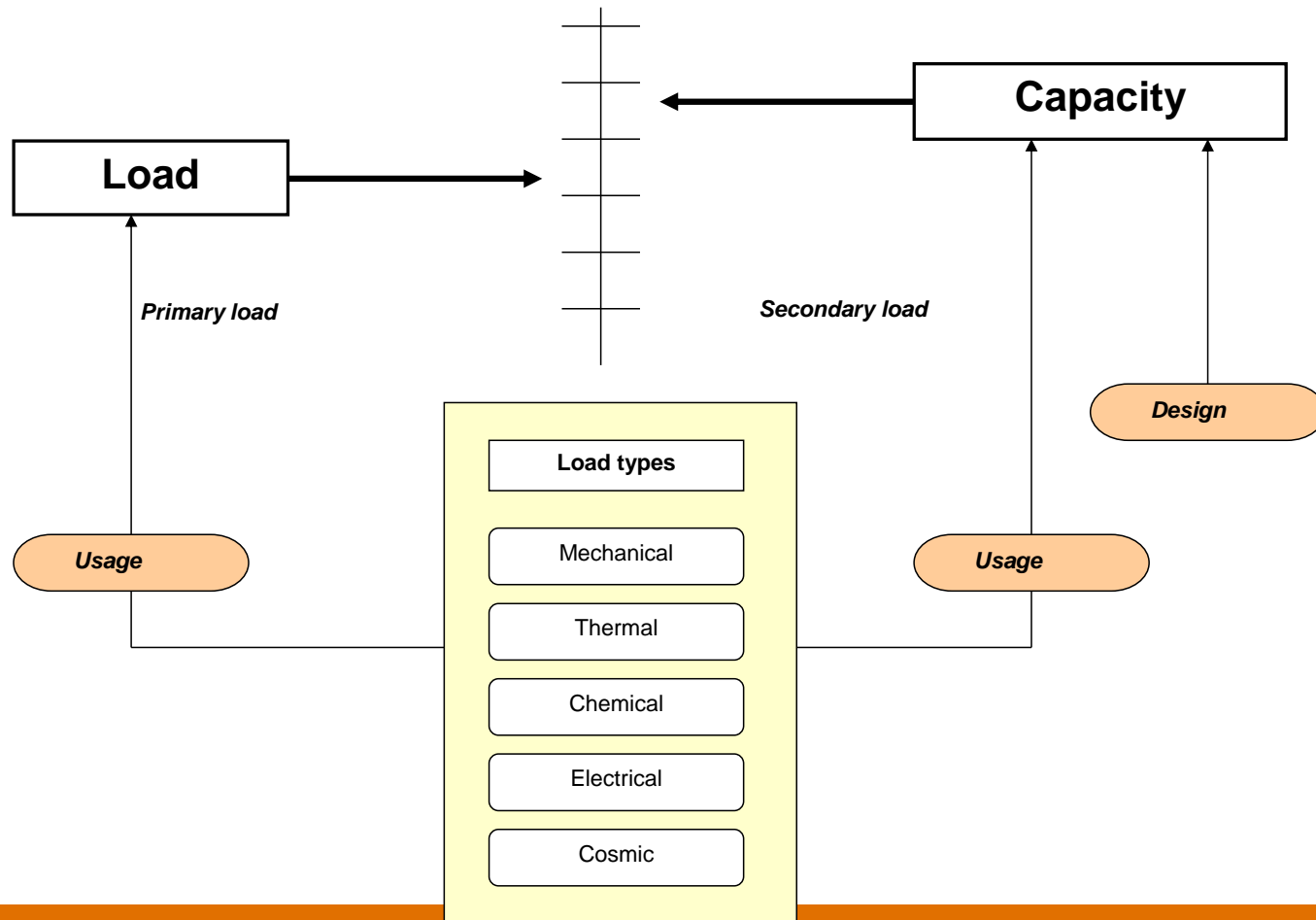
- **Preventive maintenance → length of intervals**
- **Balance between**
  - low costs → efficient
  - high availability → effective
- **Two approaches to find optimum**
  - Experience based approach
    - › predict failures based on historic data
    - › Reliability Engineering / RAMS analyses
    - › requires large datasets / inaccurate for changing conditions
  - Model based approach
    - › predict failures with physical failure models and monitored (or prescribed) variation in usage
    - › quantitative relation between usage and degradation
    - › enables *dynamic maintenance*

# Failure

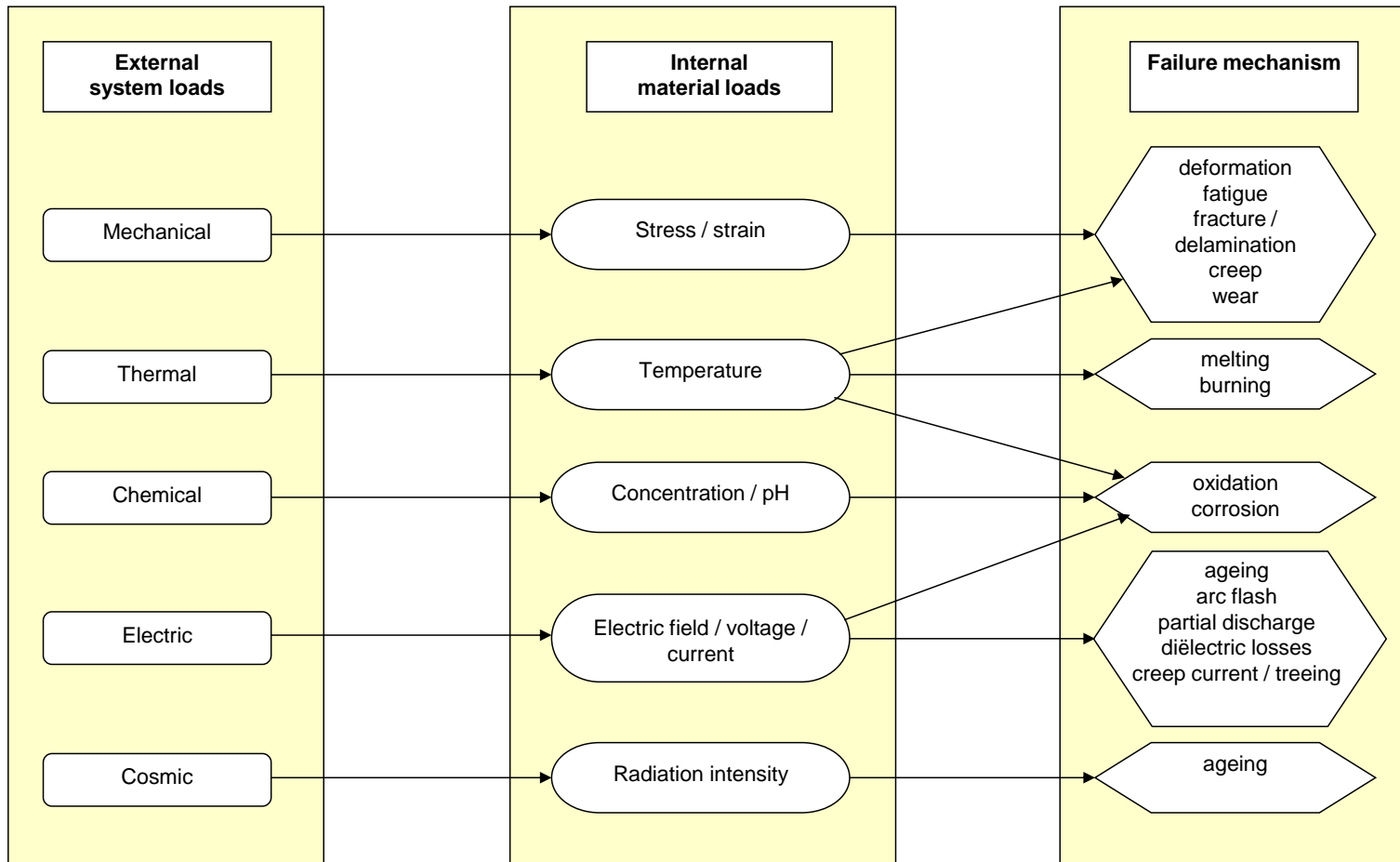
- **Requires understanding of failure behaviour**
  - How ? Why ? When ?
- **Failure: “reaching such a state that the intended function can no longer be fulfilled”**
  - depends on function
  - not always physical failure
- **Failure mode**
  - Manner in which a system functionally fails
  - Detected by decreased performance or inspection
  - Several hierarchical levels
  - Also non-physical causes: human errors / contamination
- **Failure mechanism**
  - Physical or chemical process yielding degradation and leading to failure
  - Limited number !

# Balance

- **Load versus load-carrying capacity**



# Load types and failure mechanisms

**Loads****Capacity**



## Problem 1

### Determine for the failures mentioned below

- the failure mechanism
- the governing load
- the load-carrying capacity
- a preventive solution

- **Fractured window hit by football**



- **Broken bicycle chain**





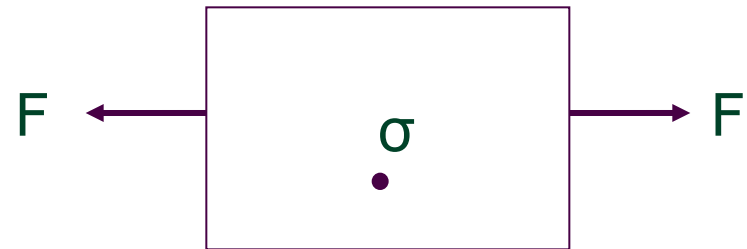
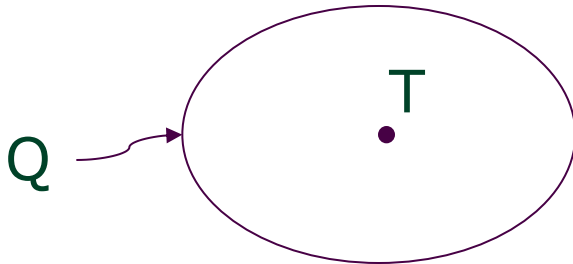
# LOADS

# Loads

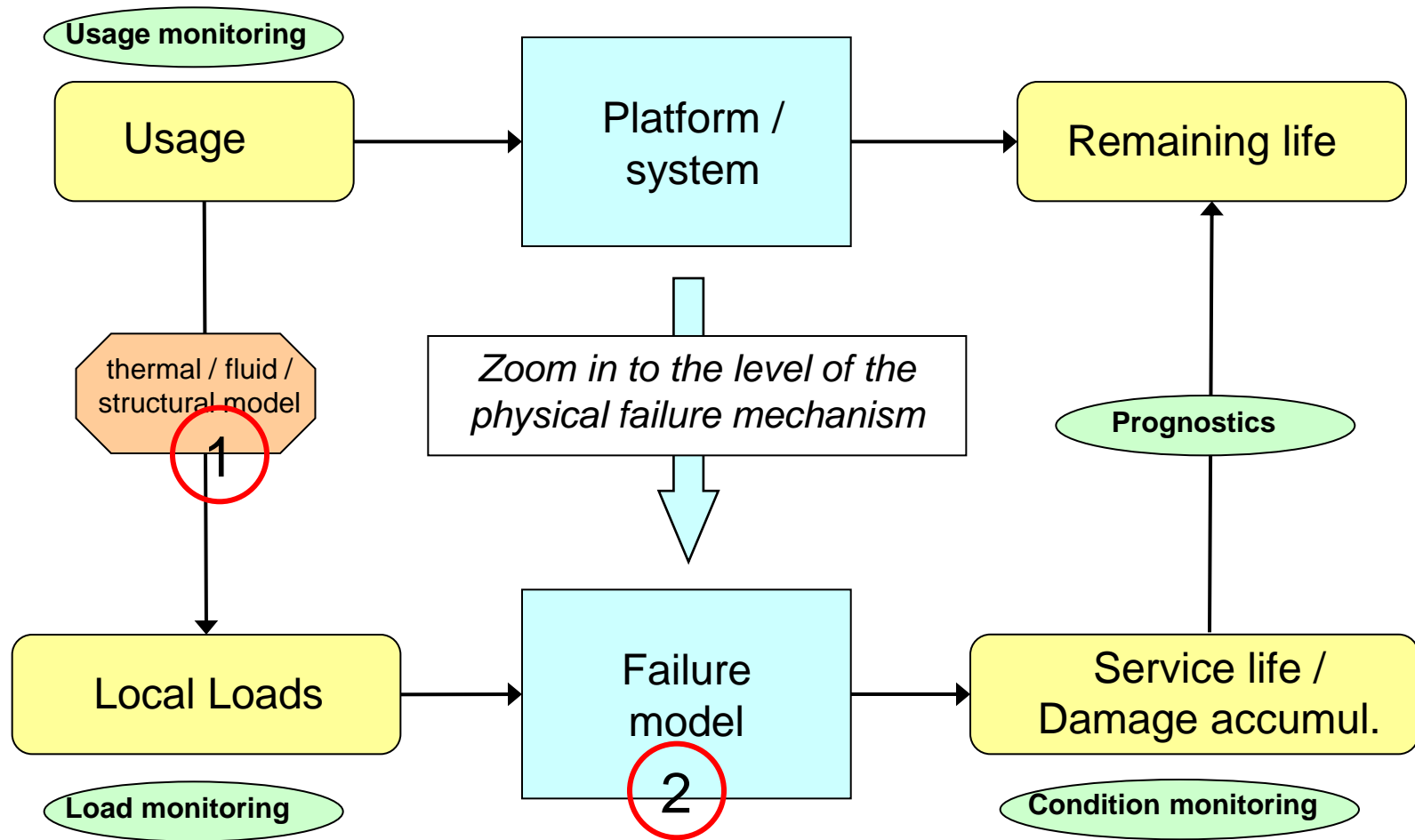
- **External loads vs internal loads**
- **Different load types**
  - Mechanical
  - Thermal
  - Electric
  - Chemical
  - Radiative
- **Relation with usage**

## External and internal loads

- **Failure occurs on (microscopic) material level**
  - mechanical fracture = release of atom bonds
  - breakdown of electric insulator = release of electrons
- **Failure is governed by local *internal* load**
  - local stress, strain, electric field, temperature
- **Operator determines *external* loads**
  - Force, moment, charge distribution, heat input

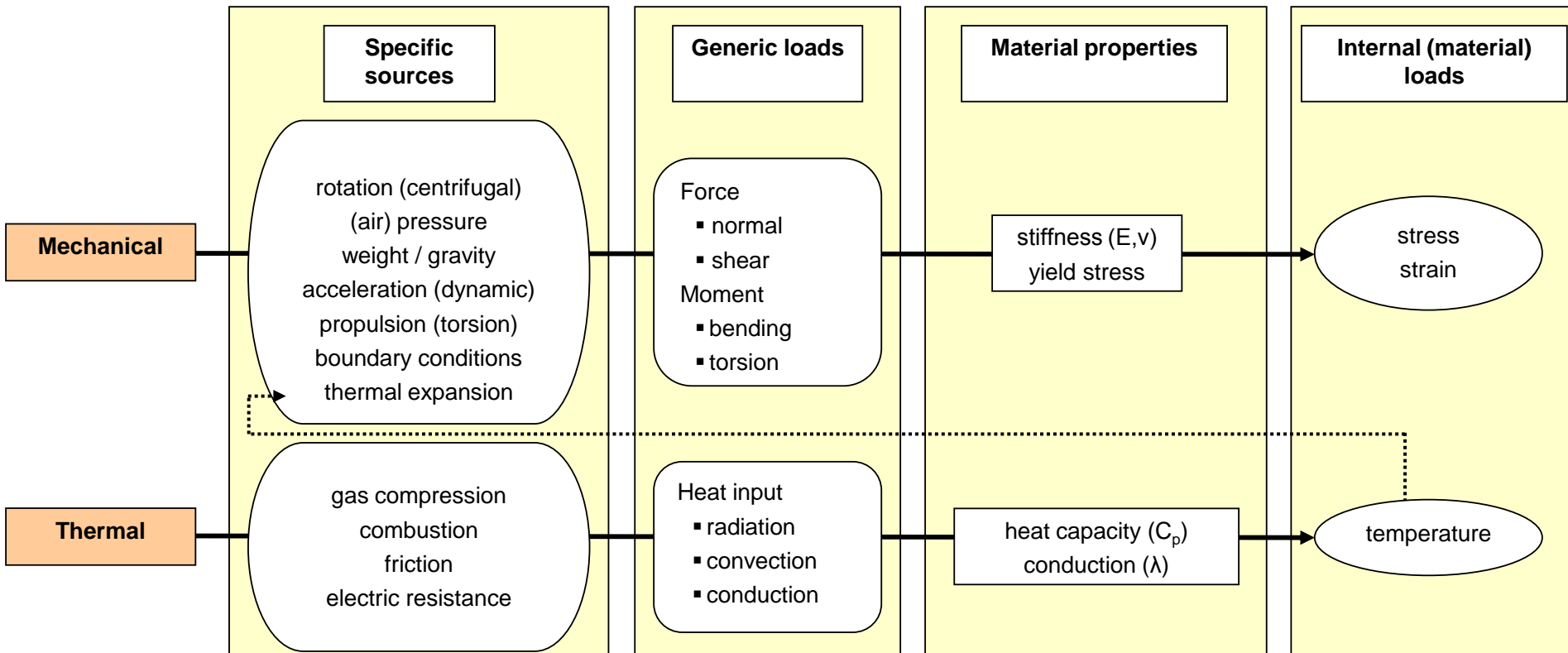


# Relating usage to service life



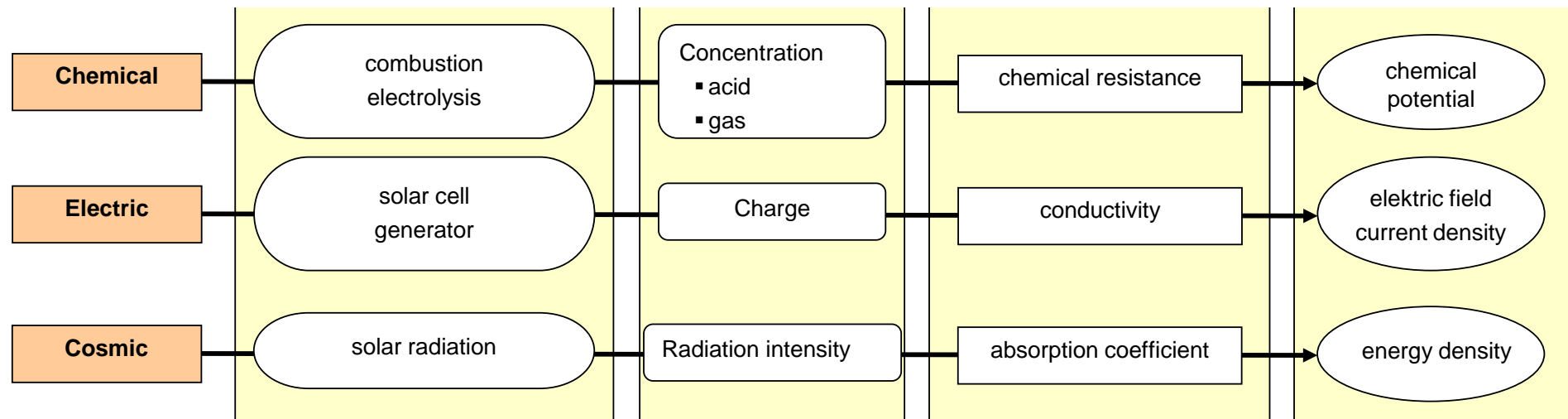
## Load types

- **Mechanical:** Forces/moments caused by various sources



- **Thermal:** Heat flow caused by several sources

## Load types (2)



# Examples of sources mechanical loads

- **Pressure**

- distributed load exerted by environment of body:
  - › water pressure (hull of ship, submarine)
  - › air / gas pressure (gas pressure vessel, aircraft wing)
- total force determined by distributed load ( $f$  in N/m<sup>2</sup>) and surface ( $A$ ):

$$F = \int_A f dA$$

- **Acceleration:**

- acceleration ( $a$ ) requires force (Newton's law) proportional to mass ( $m$ ) and  $a$ :  $F = m a$
- Rotation is acceleration → centrifugal force acts on rotating body (angular velocity  $\omega$  and radius  $r$ ):

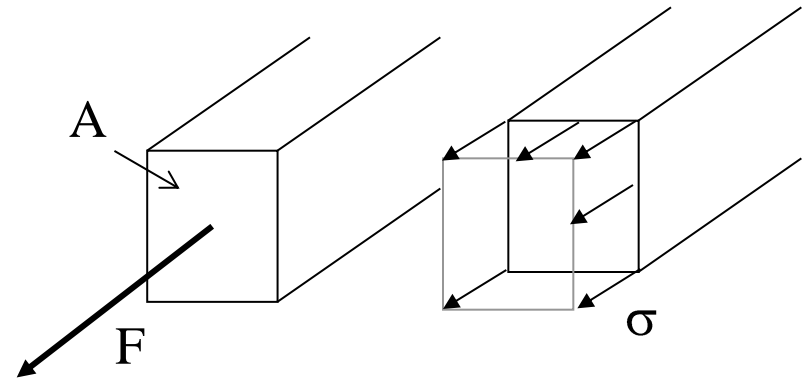
$$F_{cf} = m\omega^2 r$$



# Internal mechanical loads: stress

- **Local load parameter**
- **Relating external load (force / moment) to the properties of cross section**
- **Normal force**
  - uniform normal stress  $\sigma$

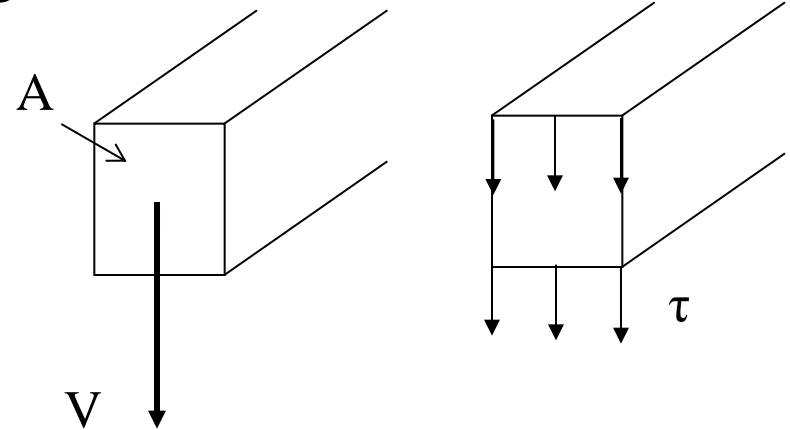
$$\sigma = \frac{F}{A}$$



## Internal mechanical loads: stress (2)

- **Similar for shear, bending and torsion**

$$\tau = \frac{V}{A}$$



$$\sigma = \frac{Md}{I_x}$$

$$I_x = \frac{1}{12}bh^3$$



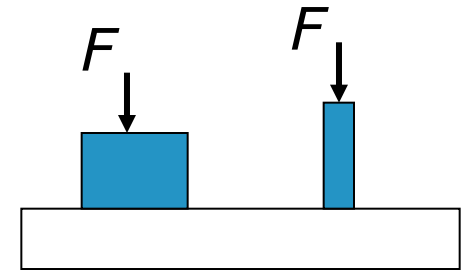
$$\tau(r) = \frac{Tr}{I_p}$$

$$I_p = \frac{\pi}{32} D^4 - d^4$$



## Thermal loads

- **Heat input per unit time and unit area:  $q$  [W/m<sup>2</sup>]**
- **internal heat generation**
  - friction  $Q = F_w v = \mu F_n v$
  - electric resistance  $Q = \frac{V^2}{R} = I^2 R$
- **external heat generation**
  - compression  $\frac{pV}{T} = \text{constant}$
  - combustion
- **causes increase or decrease of temperature:  
 $T$  [°C of K]**



## Internal load: temperature

- **temperature is state variable: always a value**
- **only changes when (nett) amount of heat is added / removed**
- **temperature increase depends on amount of heat  $Q$  [J], mass and material:**

$$\Delta T = \frac{Q}{m c_p}$$

- **Typical values  $c_p$** 
  - metal: 100 – 500 J/kgK
  - water: 4000 J/kgK

## Summary

- **External loads**
  - Type and magnitude determined by usage of system
- **Internal loads**
  - Depend on magnitude external load + properties / dimensions
  - Directly responsible for failure



- **Usage + properties determine loads**
- **Compare with capacity → failure mechanisms**

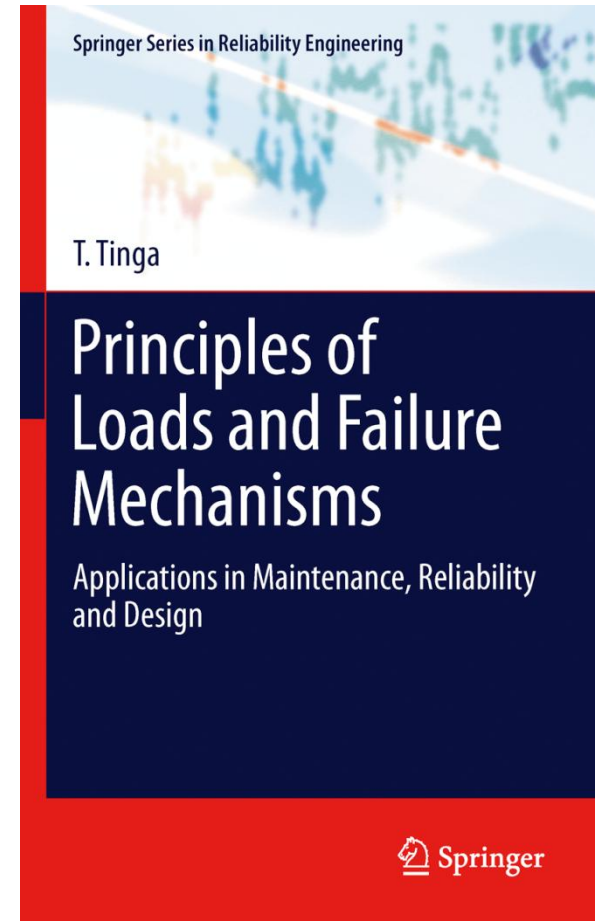


# FAILURE MECHANISMS

# Failure mechanisms

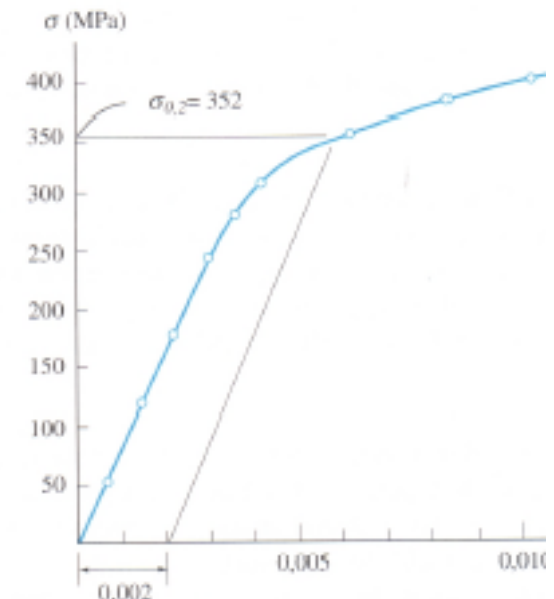
- ***Static overload***
- **Deformation**
- ***Fatigue***
- **Creep**
- ***Wear***
- **Melting**
- **Thermal degradation**
- **Electric failures**
- **Corrosion**
- **Radiative failures**

- **Complete overview:**



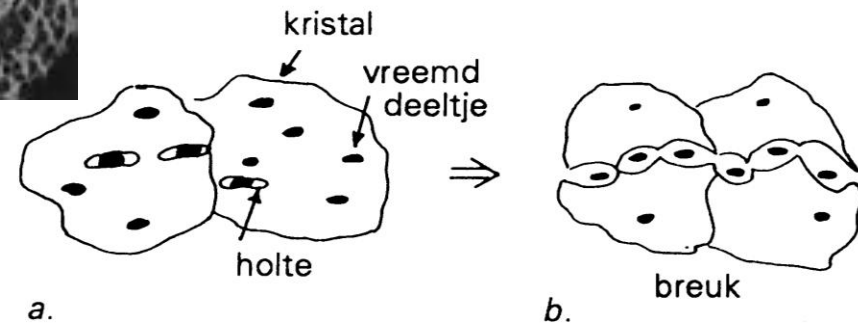
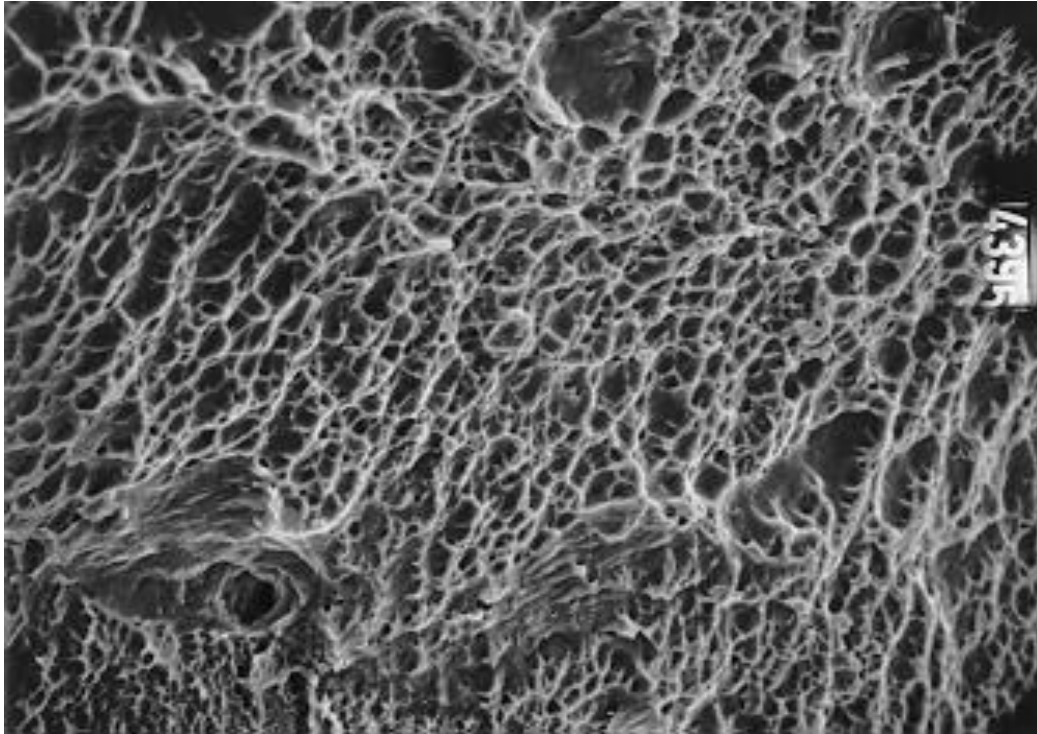
# Static overload

- **when stress > strength**
  - *load* = stress
  - *capacity* = material strength
- **tensile strength determined by tensile test**
- **tensile strength temperature dependent**
  - $T \uparrow \quad \sigma_t \downarrow$
- **design: expected load < strength**
  - effect temperature
  - safety factors
- **Failure analysis**
  - fracture surface → 'dimples'



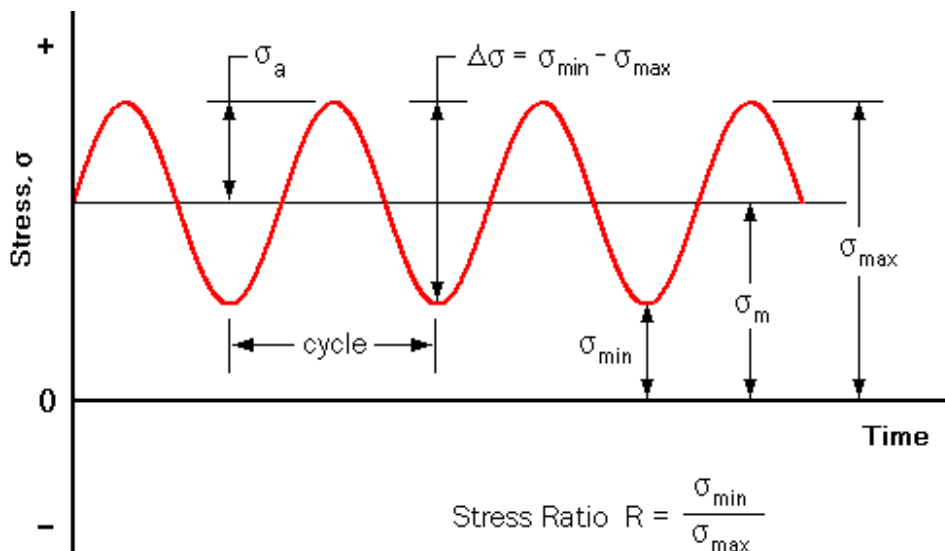


# Fracture surface static overload



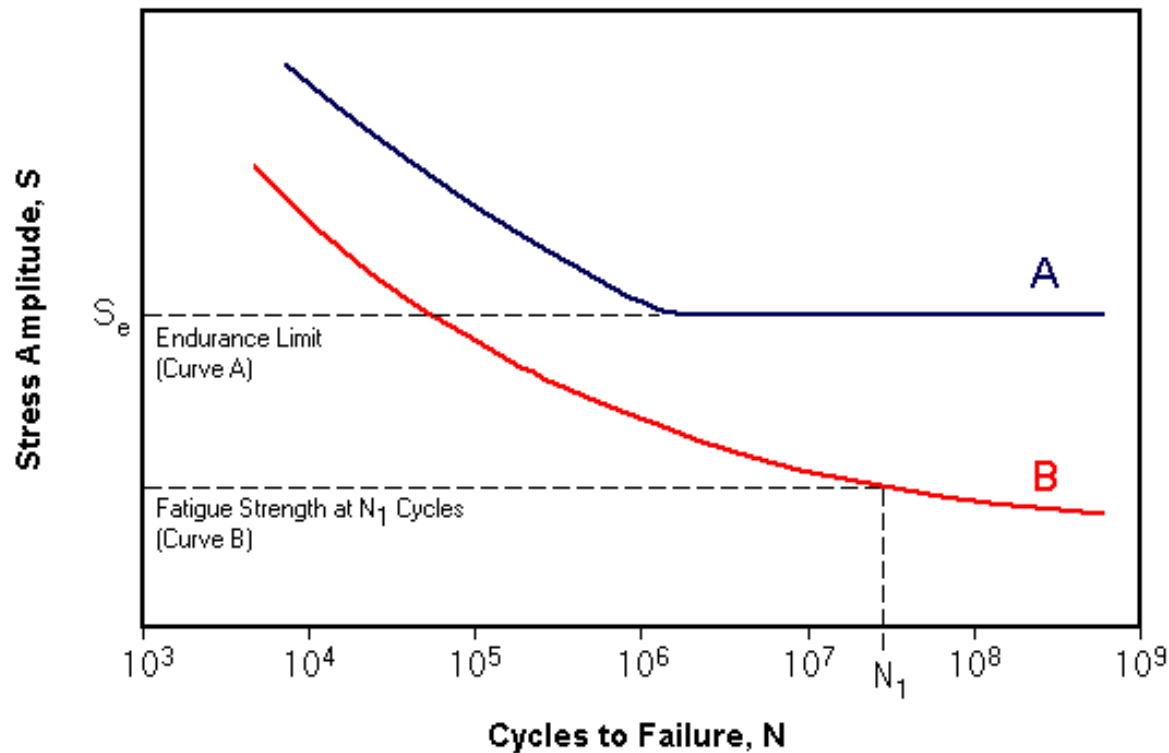
# Fatigue

- **caused by cyclic load**
- **stress level below tensile strength**
- **failure occurs after large number of cycles ( $10^4 - 10^7$ )**
- ***load* = strain or stress cycle ( $\Delta\varepsilon$  or  $\Delta\sigma$ )**

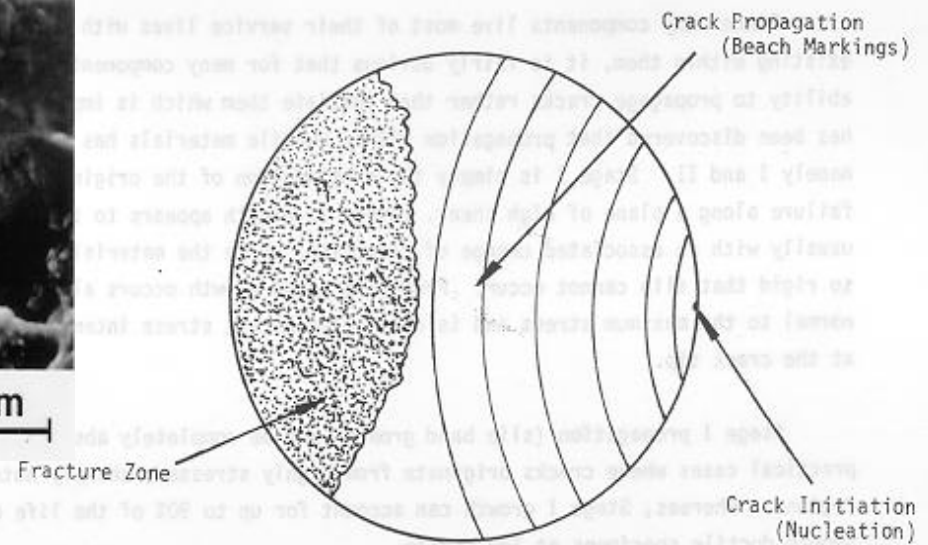
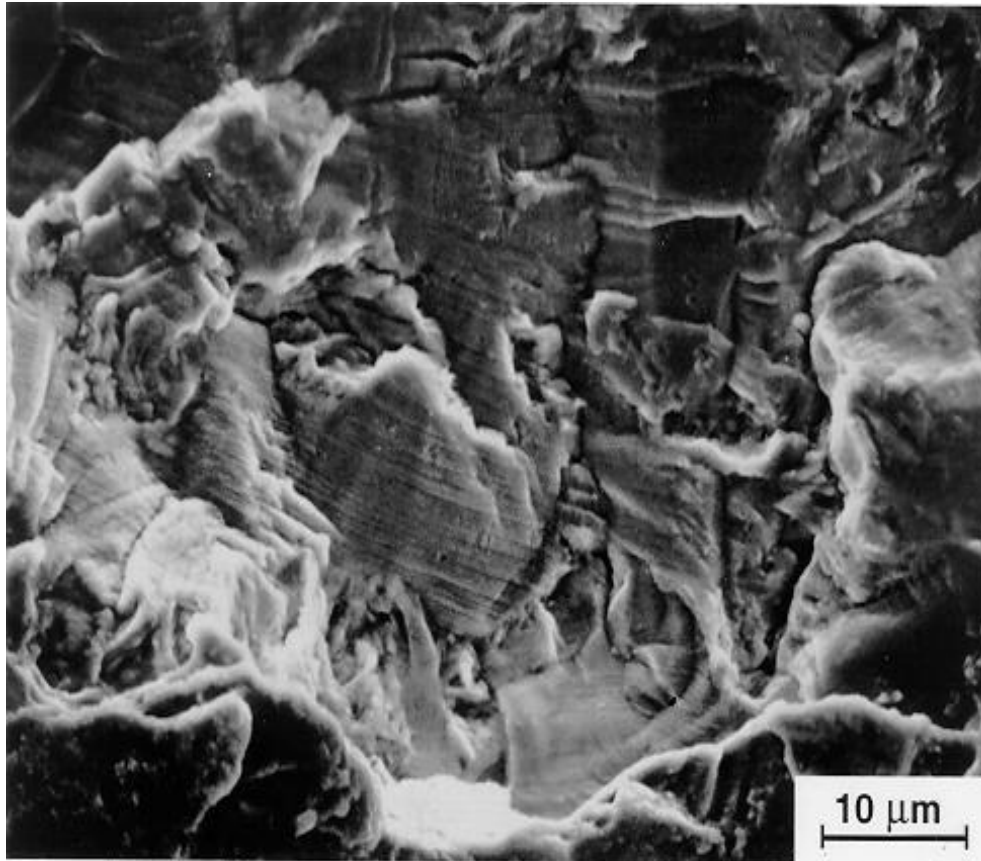


## Fatigue (2)

- **Capacity = fatigue resistance**
  - S-N diagram or Wöhler-curve



# Failure analysis



## Life assessment

- **Constant amplitude: from  $S-N$  or Smith diagram**
- **Variable amplitude: Miner damage rule**
  - $n$  cycles at load with life  $N$  yields damage  $D$

$$D = \frac{n}{N}$$

- $0 < D < 1$  : percentage of life consumed
- $p$  blocks of  $n_i$  cycles with life  $N_i$  yields damage

$$D = \sum_{i=1}^p D_i = \sum_{i=1}^p \frac{n_i}{N_i}$$

- no sequence effects

## Problem 2

For a gas turbine blade it has been derived:

- $\sigma = 3.95 N^2$

A start/stop cycle is defined as:

- start/stop: N from 0 to 14.000 rpm and back to 0

- **Calculate the number of start/stops before the blade fails due to fatigue**



In addition to the start/stop cycle two more cycles exist:

- manoeuvre: N from 8.000 to 14.000 rpm and back to 8.000 rpm
- correction: N from 10.000 to 12.000 and back to 10.000 rpm

The usage profile of the engine is: per flight 1 start/stop, 3 manoeuvres and 10 corrections.

- **Calculate the number of flights to failure**

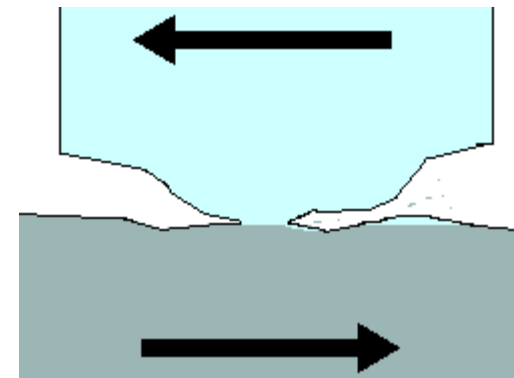
# Wear

- **occurs when parts move relative to each other or along liquid / gas**
- **two types**
  - two-body wear mechanisms
    - » on contact between two parts
  - single-body wear mechanisms
    - » on surface due to flowing medium
- **resulting in**
  - bad fitting
  - vibrations (e.g. bearings)
  - cracks – fracture

# Wear mechanisms

## Two-body wear

- **abrasive wear**
- **corrosive wear**
- **surface wear**
- **adhesive wear**
  - strong bonding between peaks of surface roughness
  - high friction → high temperature
  - improve by lowering friction:
    - » materials
    - » lubrication



## Single body wear

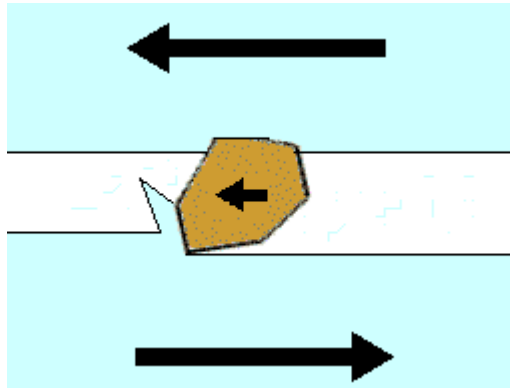
- **Erosion**



## Wear mechanisms (2)

- **abrasive wear**

- when considerable differences in hardness ( $> 20\%$ )
- presence of hard particles
  - » as additional body (three-body): sand
  - » fractured particles (debris)

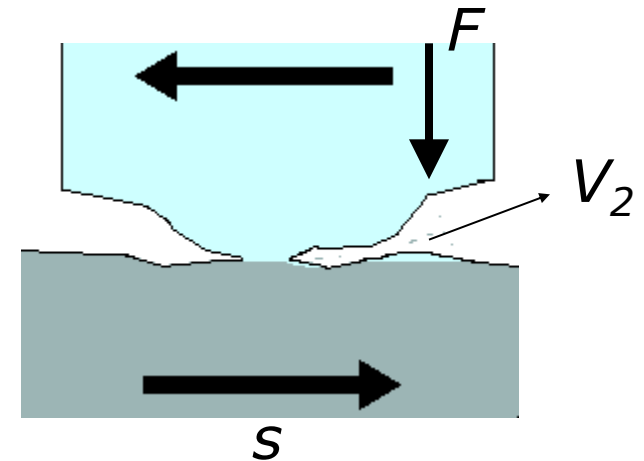


# Wear rate

- **Archard's law**

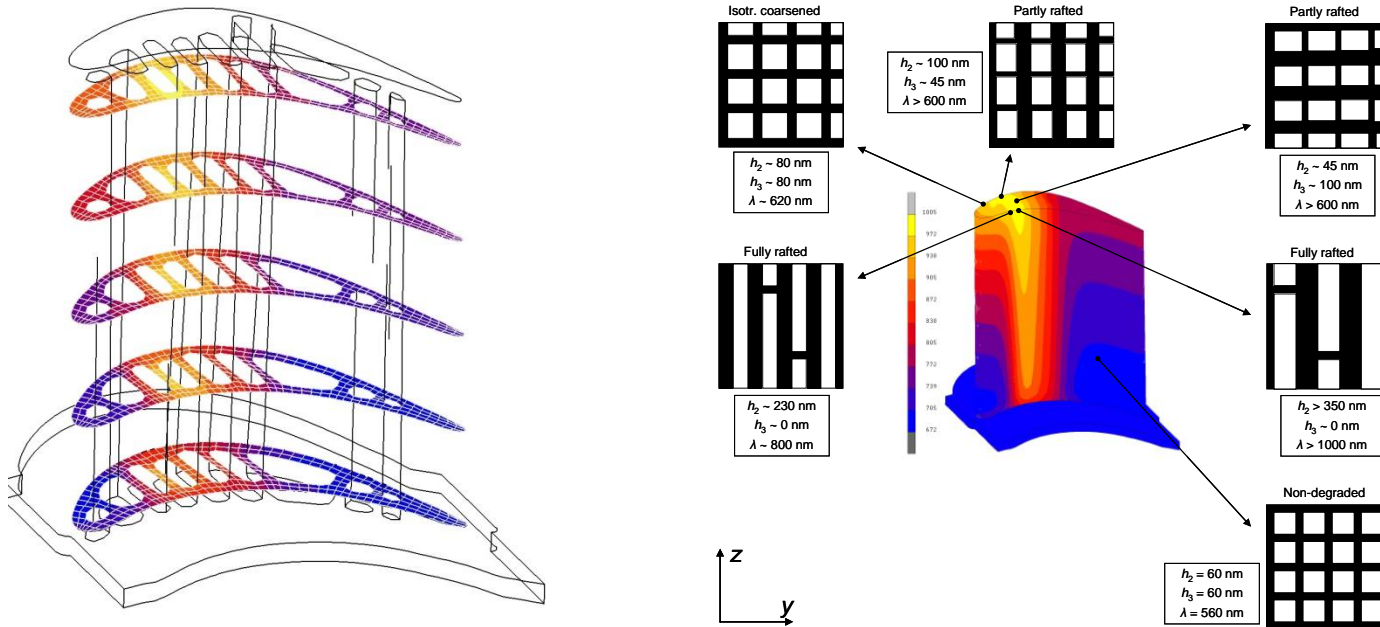
$$V_i = k_i F s$$

- $k$  [mm<sup>3</sup>/Nm] is specific wear rate (different for two bodies)
- $k$  depends on
  - » material combination
  - » surface roughness
  - » contact temperature
  - » hardness
  - » lubrication



## More advanced methods

- Calculate loads with FE / CFD models
- Complex (multi-scale) damage models



- Find balance between effort and benefits !

## Summary

- **Loads – capacity balance**
- **Load types**
- **Failure mechanisms**



- **Application in maintenance**
  - Preventive maintenance and prognostics
  - RAMS analysis → Relevant Failure Parameter
  - Condition monitoring
  - Root cause analysis



# APPLICATIONS

# Application in maintenance

## Knowledge on failure (mechanisms) can be used ...

- **before failures occur**
  - Identify critical components → FMECA
  - Predict time to failure → determine optimal maintenance intervals
  - Develop efficient condition monitoring systems
- **after failure has occurred**
  - Why did component fail ?
  - How can future failures be prevented ?
  - Root Cause Analysis
- **when a fraction of a (larger) population has failed**
  - Quantify failure behaviour
  - Find Relevant Failure Parameter (RFP)

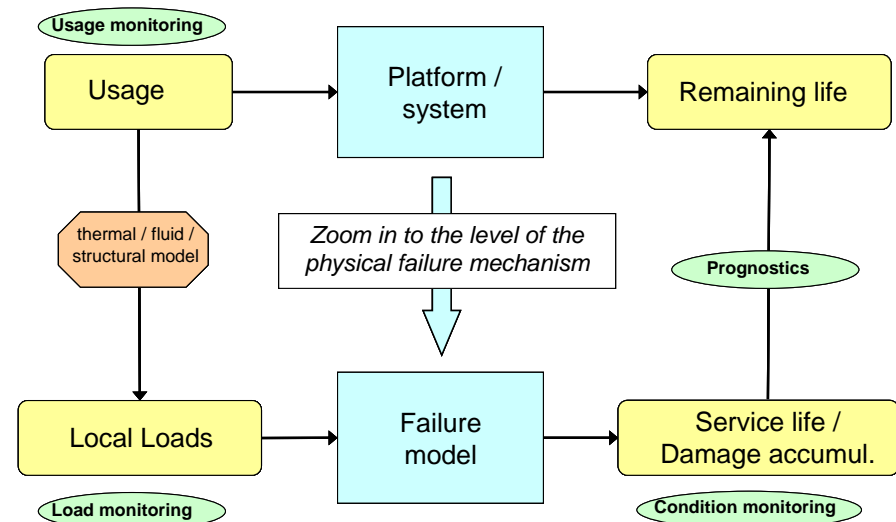
# 1. Preventive maintenance & prognostics

## • Procedure

- Monitor actual usage of the system
- Derive internal loads from the usage
- Develop physical models for critical failure mechanism(s)
- Calculate service life consumption and remaining life
- See *Tinga, Reliability Engineering and System Safety 2010*

## • Applied to several military systems

- Chinook helicopter
- NH-90 helicopter
- CV-90 combat vehicle
- LC frigate



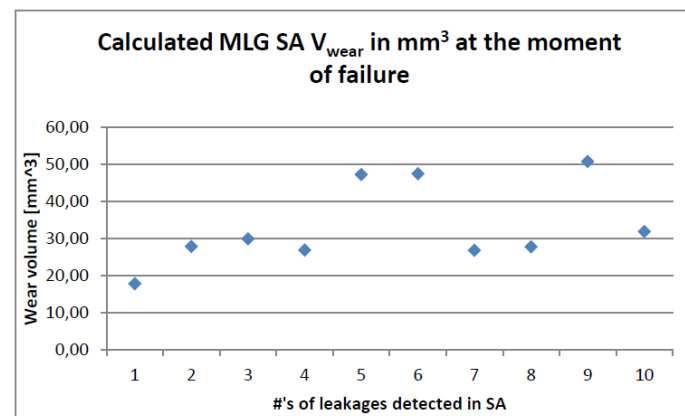
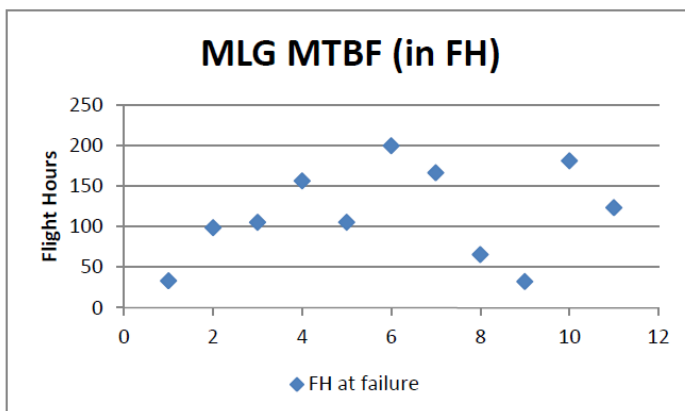
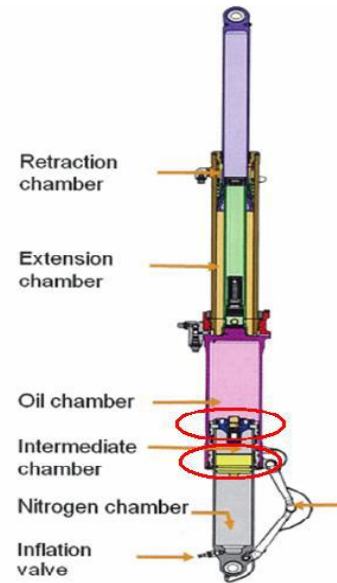
## NH-90 helicopter prognostics

- **Identified critical components**
  - Cost drivers
  - Availability killers
- **Determined failure mechanism + governing loads**
- **HUMS system available for monitoring**
  - Usage → flight hours, landings, conditions, etc.
  - Health → mainly vibrations
- **Maintenance primarily related to flight hours**



## NH-90 helicopter prognostics (2)

- **For some components better parameters**
  - Landing gear shock absorber
    - › Mechanism: wear of seal
    - › Usage parameter: travelled distance  
→ # landings + weight
  - Blade folding system locking mechanism
    - › Mechanism: corrosion
    - › Usage parameter: *saline* flight hours
  - See Heerink, TU Delft 2012

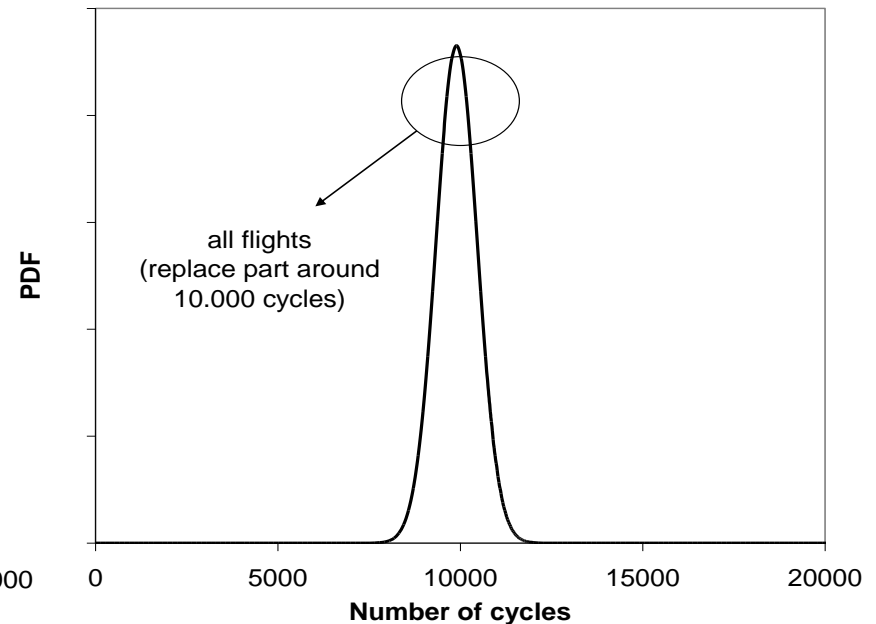
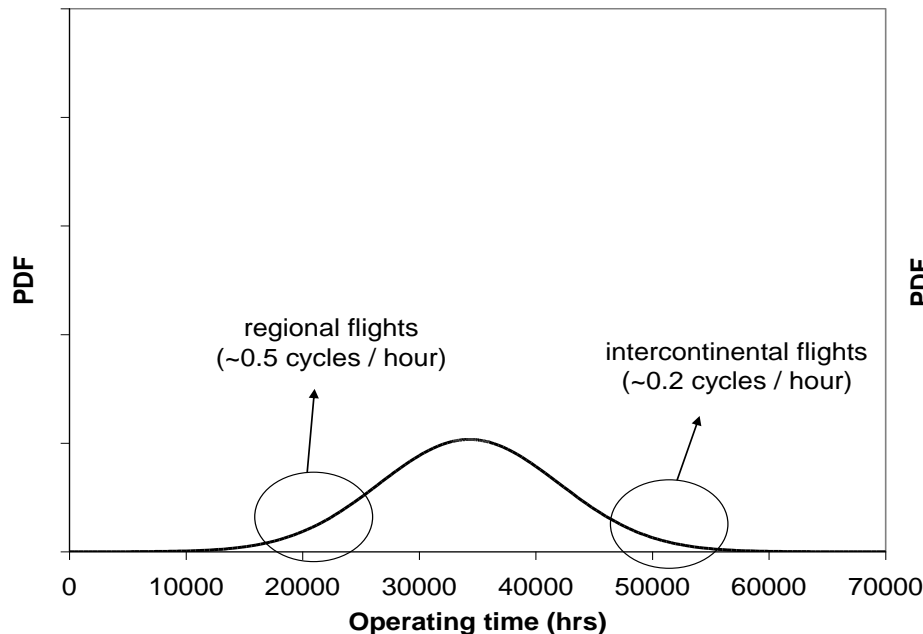


## 2. Relevant Failure Parameter

- **RAMS / data analyses → typical approach**
  - Collect failure data - *t<sub>tf</sub>* in calendar / operating time
  - Analyze data
    - › Mean Time Between Failures (MTBF)
    - › Parameters of distribution function (Weibull, exponential)
  - Adapt maintenance policy to obtained values
- **Problem: uncertainty / variation**
  - Variation in usage / conditions causes variation in MTBF
  - Ineffective or inefficient maintenance
- **Solution**
  - Determine more relevant failure parameter
    - › Operating hours, starts, cycles, ... in stead of hours / days

## Relevant Failure Parameter - example

- **Airliner with fleet of aircraft**
- **Part fails due to fatigue ( $\sim 10.000$  cycles)**



## Problem 3 - RFP

- **A car is maintained every 15.000 km / 1 year**
  - Many parts replaced or inspected
  - Usage parameter: km or calendar time
- **What would be the most relevant failure parameter for:**
  1. brake discs
  2. tires
  3. structure
  4. engine lubrication oil
  5. hydraulic oil in the brake system
- **Is usage based maintenance the most appropriate policy ?**



### 3. Condition monitoring

- **Sensing and data acquisition now well developed**
- **Remaining challenges**
  1. Translation of data into maintenance information
    - only when condition is not directly monitored
      - performance
      - usage
  2. Add prognosis to diagnosis
  3. Selection of proper measurements and location

**Condition Monitoring → Condition Based Maintenance**

## Diagnosis vs. Prognosis

- **Condition monitoring → assess the present condition**
- **Need for *prediction* of future maintenance**
  - › less risk / better planning
- **Two options:**
  - wait for indication of failure / degradation (*diagnostic*)
    - » often based on certain threshold value with safety factor
  - predict remaining life (*prognostic*)
    - » from every state prediction of expected maintenance
    - » prediction improves when reaching end of life
    - » based on assumed usage
- **Prognostic approaches**
  - raw data / statistics → trending
  - physical processes

# Development of new CM / CBM systems

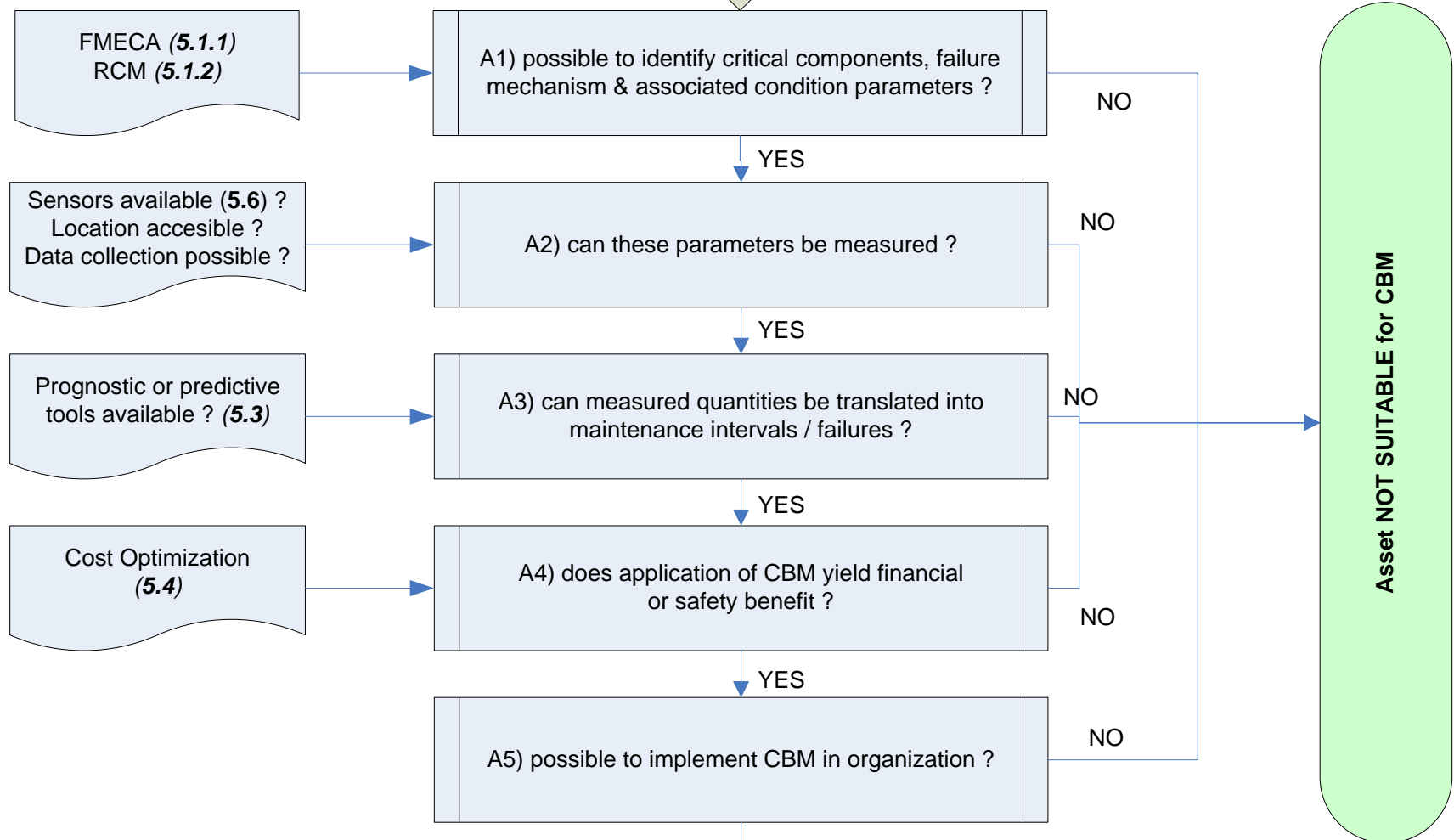
- **Selecting appropriate quantities for CM**
  - which parameters should be measured ?
  - at what location ?
  - how can measured quantities be translated into condition information ?
- **Criteria**
  - representative (knowledge on failure mechanisms)
  - measurable (sensors, accesibility)



**Condition Monitoring → Condition Based Maintenance**

## Decision scheme Condition Based Maintenance

### Information (section)



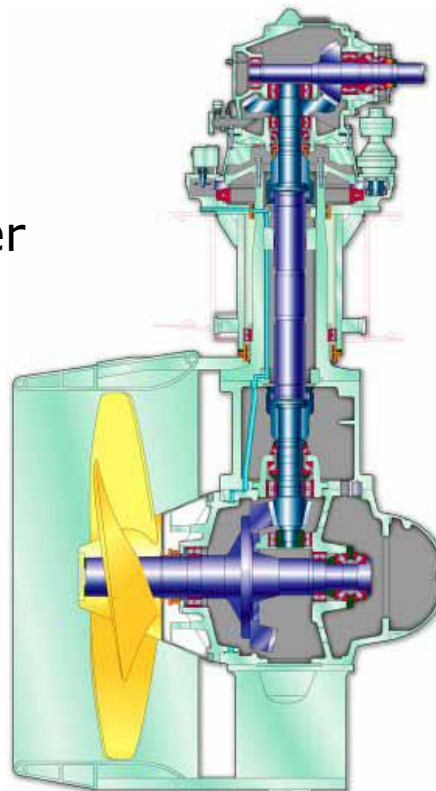
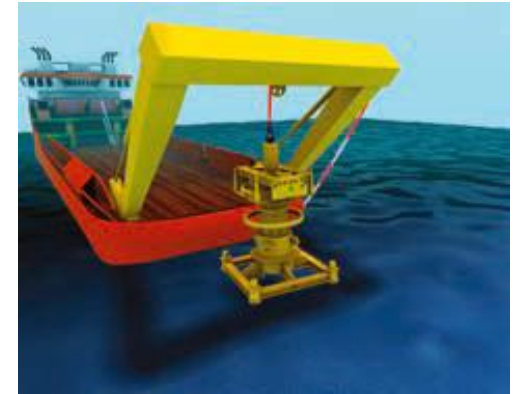


## 4. Root Cause Analysis (RCA)

- **Structured method to find the cause of a failure in a system**
- **Solving problems is often only solving the symptoms**
  - Failures keep returning
  - Low availability / high costs
- **RCA should be executed to a sufficiently deep level → failure mechanism + loads !**
- **Solution is often rather simple**
  - Reduce loads on the system
  - Increase capacity of the system

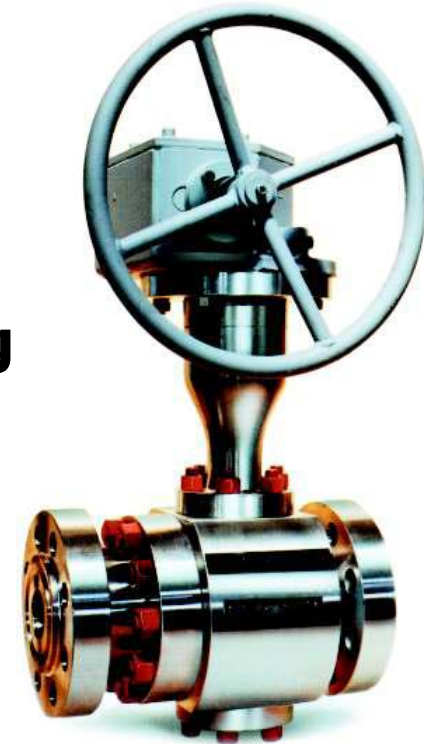
## Case studies in WCM-IP

- **Bosch Rexroth**
  - hydraulic cylinder for off-shore
- **Wärtsilä**
  - Bearing in thruster



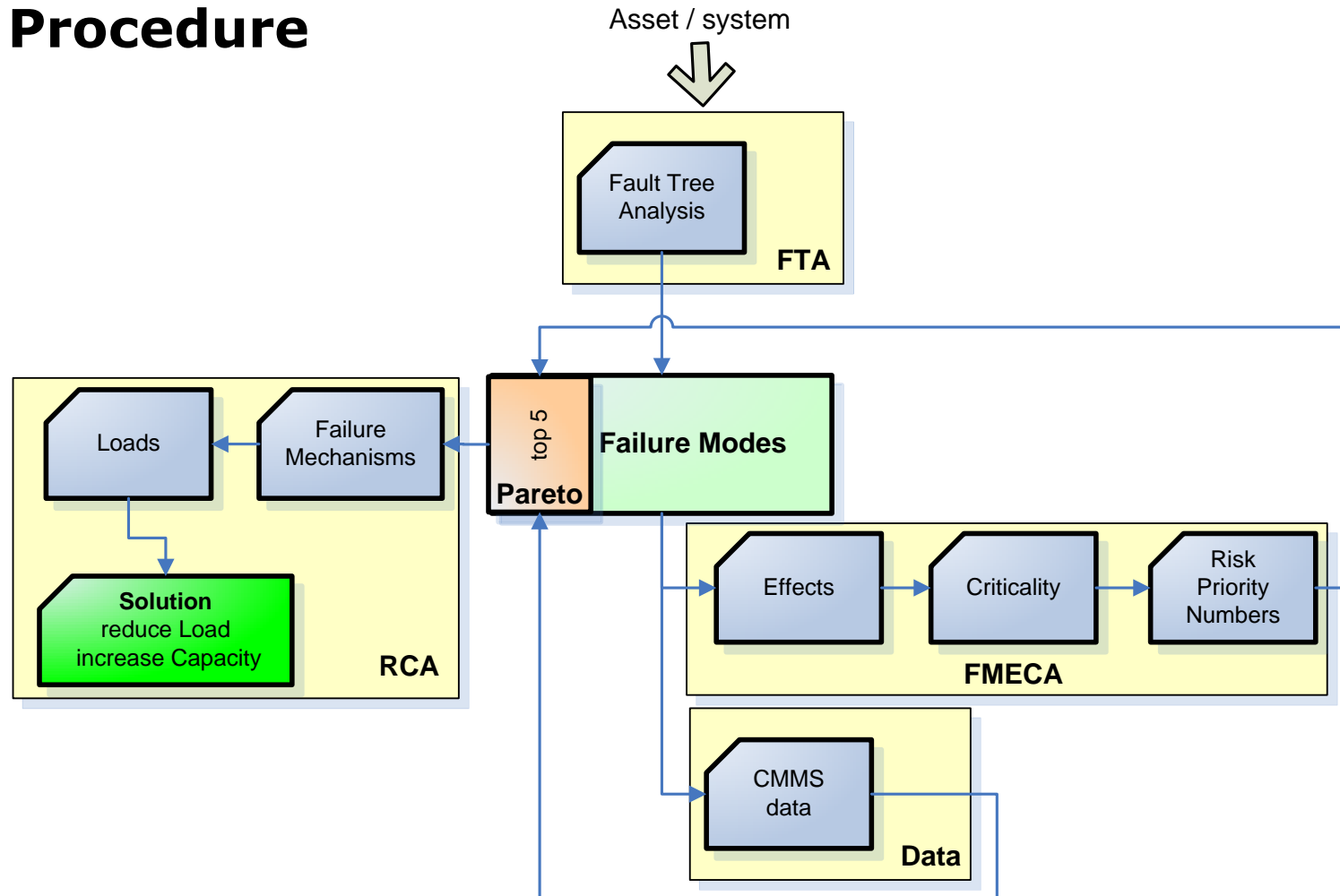
## Case studies in WCM-IP

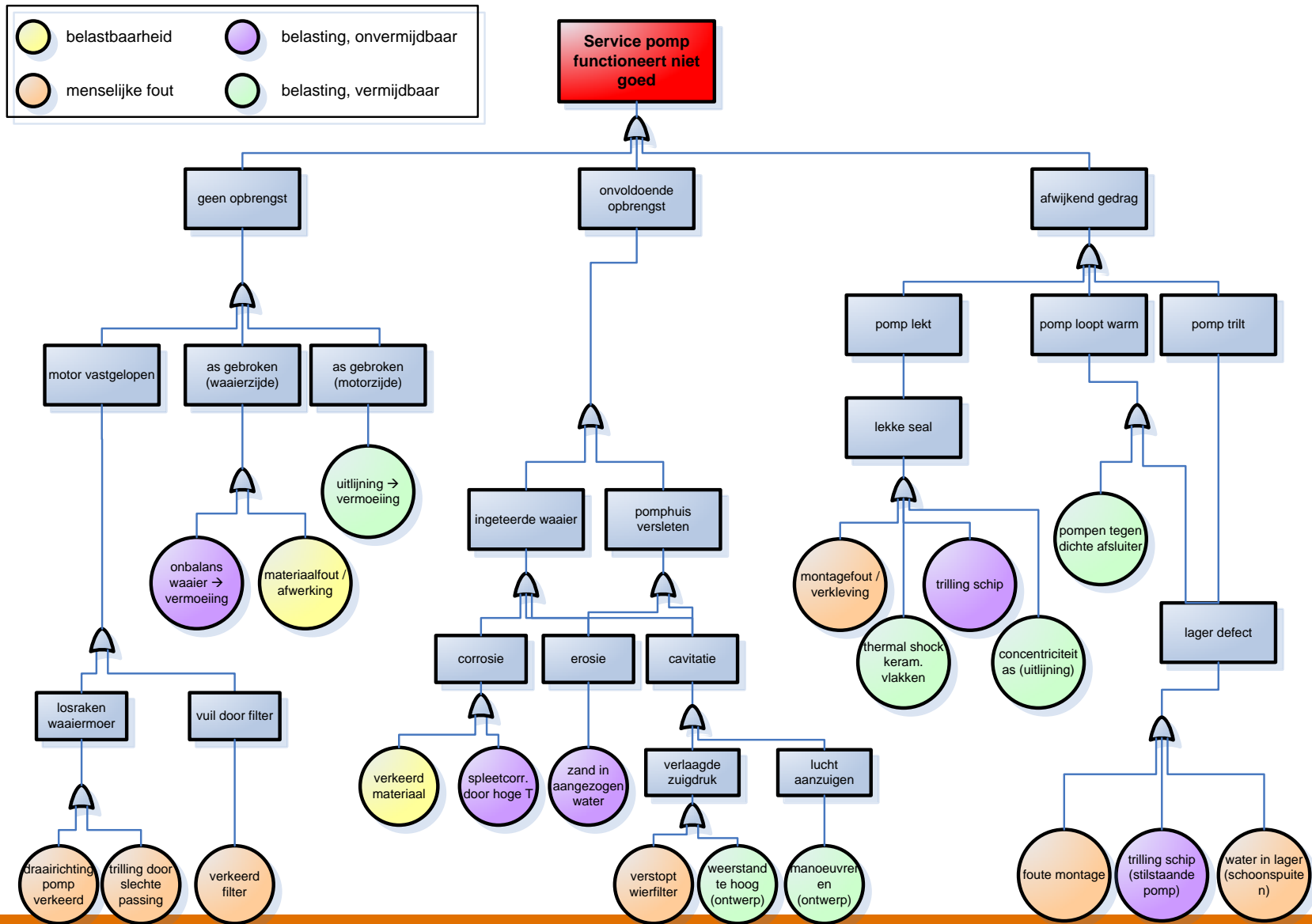
- **Gasunie → valves in gas network**
- **Koninklijke Marine → fire extinguishing pump**



# Mechanism based Failure Analysis

- **Procedure**





## 4 types of causes

### 1. Capacity too low → quality control

- e.g. wrong impeller material → corrosion

### 2. Human error → (obeying) regulations

- e.g. wrong rotation direction

### 3. Load *avoidably* too high (misuse)

- e.g. shaft fracture due to misalignment, operating with closed valve

### 4. Load *unavoidably* too high

- e.g. failure of non-rotating bearing due to vibrations

# Mechanisms & Solutions

- **Seal leakage**

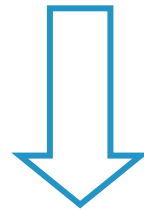
- Mechanisms
  - › wear (no water in pump during operation)
  - › thermal shock (cooling of heated seals causes fracture)
  - › damage due to vibration overload (cavitation, misalignment)
- Solution
  - › mainly caused by operating the pump incorrectly
  - › prevented by better instruction and training of the operators

- **Insufficient yield**

- Mechanisms: damaged impeller due to
  - › Corrosion, erosion by sand (shallow water), fatigue (cavit.)
- Solution
  - › most impeller failures unavoidable → due to regular usage
  - › Prediction may be possible → monitor operating hours in shallow water

## Conclusion

- **Understanding the failure behaviour enables more efficient maintenance strategies**
- **(in)balance between load and capacity is key to failures**
- **Effect of variations in usage is often neglected**



- **Model-based approach good addition to traditional experience-based approach**